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How to Determine Spiral Bevel Gear Tooth Geometry for Finite Element Analysis

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ABSTRACT

An analytical method has been developed to determine gear tooth surface coordinates of face-milled spiral bevel gears. The method combines the basic gear design parameters with the kinematical aspects for spiral bevel gear manufacturing. A computer program was developed to calculate the surface coordinates. From this data a three-dimensional model for finite element analysis can be determined. Development of the modeling method and an example case are presented.

INTRODUCTION

Spiral bevel gears are currently used in all helicopter power transmission systems. This type of gear is required to turn the corner from a horizontal engine to the vertical rotor shaft. These gears carry large loads and operate at high rotational speeds. Recent research has focused on understanding many aspects of spiral bevel gear operation, including gear geometry [1-12], gear dynamics [13-15], lubrication [16], stress analysis and measurement [17-21], misalignment [22,23], and coordinate measurements [24,25], as well as other areas.

Research in gear geometry has concentrated on understanding the meshing action of spiral bevel gears [8-11]. This meshing action often results in much vibration and noise due to an inherent lack of conjugation. Vibration studies [26] have shown that in the frequency spectrum of an entire helicopter transmission, the highest response can be that from the spiral bevel gear mesh. Therefore if noise reduction techniques are to be implemented effectively, the meshing action of spiral bevel gears must be understood.

Also, investigators [18,19] have found that typical design stress indices for spiral bevel gears can be significantly different from those measured experimentally. In addition to making the design process one of trial and error (forcing one to rely on past experience), this inconsistency makes extrapolating over a wide range of sizes difficult, and an overly conservative design can result.

The objective of the research reported herein was to develop a method for calculating spiral

bevel gear-tooth surface coordinates and a three-dimensional model for finite element analysis. Accomplishment of this task requires a basic understanding of the gear manufacturing process, which is described herein by use of differential geometry techniques [1]. Both the manufacturing machine settings and the basic gear design data were used in a numerical analysis procedure that yielded the tooth surface coordinates. After the tooth surfaces (drive and coast sides) were described, a three-dimensional model for the tooth was assembled. A computer program was developed to automate the calculation of the tooth surface coordinates, and hence, the coordinate for the gear-tooth three-dimensional finite element model. The basic development of the analytical model is explained, and an example of the finite element method is presented.

DETERMINATION OF TOOTH SURFACE COORDINATES

The spiral gear machining process described in this paper is that of the face-milled type. Spiral bevel gears manufactured in this way are used extensively in aerospace power transmissions (i.e., helicopter main/tail rotor transmissions) to transmit power between horizontal gas turbine engines and the vertical rotor shaft. Because spiral bevel gears can accommodate various shaft orientations, they allow greater freedom for overall aircraft layout.

In the following sections the method of determining gear-tooth surface coordinates will be described. The manufacturing process must first be understood and then analytically described. Equations must be developed that relate machine and workpiece motions and settings with the basic gear design data. The simultaneous solution of these equations must be done numerically since no closed-form solution exists. A description of this procedure follows.

Gear Manufacture

Spiral bevel gears are manufactured on a machine like the one shown in Fig. 1. This machine cuts away the material between the concave and convex tooth surfaces of adjacent teeth simultaneously. The machining process is better illustrated in Fig. 2. The head cutter (holding the cutting blades or the grinding wheel) rotates about its own axis at the proper cutting speed,

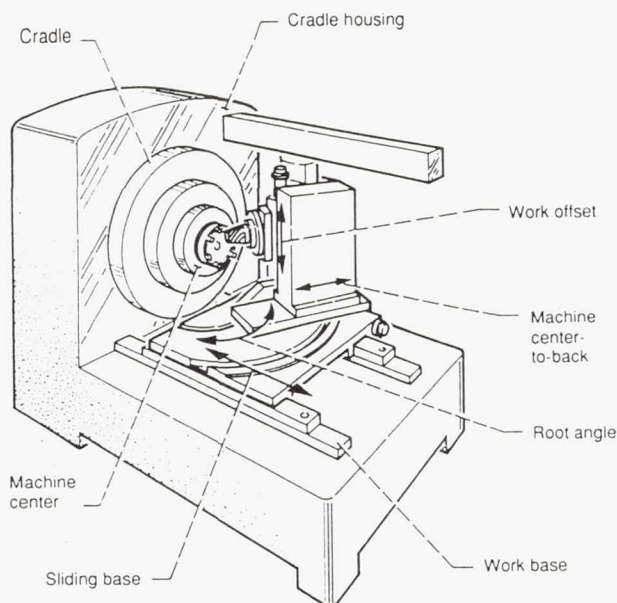


Figure 1.—Machine used for spiral bevel gear tooth surface generation.

about its own axis at the proper cutting speed, independent of the cradle or workpiece rotation. The head cutter is connected to the cradle through an eccentric that allows adjustment of the axial distance between the cutter center and cradle (machine) center, and adjustment of the angular position between the two axes to provide the desired mean spiral angle. The cradle and workpiece are connected through a system of gears and shafts, which controls the ratio of rotational motion between the two (ratio of roll). For cutting, the ratio is constant, but for grinding, it is a variable.

Computer numerical controlled (CNC) versions of the cutting and grinding manufacturing processes are currently being developed. The basic kinematics, however, are still maintained for the generation process; this is accomplished by the CNC machinery duplicating the generating motion through point-to-point control of the machining surface and location of the workpiece.

Coordinate Transformations

The surface of a generated gear is an envelope to the family of surfaces of the head cutter. In simple terms this means that the points on the generated tooth surface are points of tangency to the cutter surface during manufacture. The conditions necessary for envelope existence are given kinematically by the equation of meshing. This equation can be stated as follows: the normal of the generating surface must be perpendicular to the relative velocity between the cutter and the gear-tooth surface at the point in question [1].

The coordinate transformation procedure that will now be described is required to locate any point from the head cutter into a coordinate system rigidly attached to the gear being manufactured. Homogeneous coordinates are used to

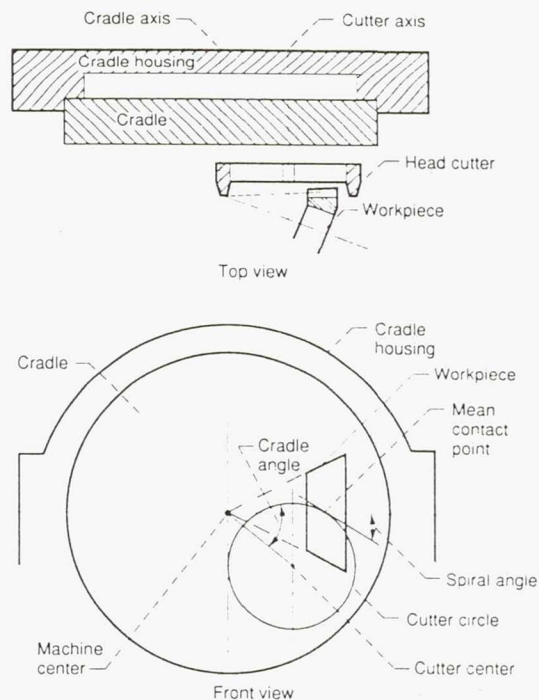


Figure 2.—Orientation of workpiece to generation machinery.

allow rotation and translation of vectors simply by multiplying the matrix transformations. The method used for the coordinate transformation can be found in Refs. 1, 5, 8 to 11, and 27.

To begin the process we start with the head cutter (Fig. (3)). In this report it is assumed that the cutters are straight-sided. The parameters u and θ determine the location of a

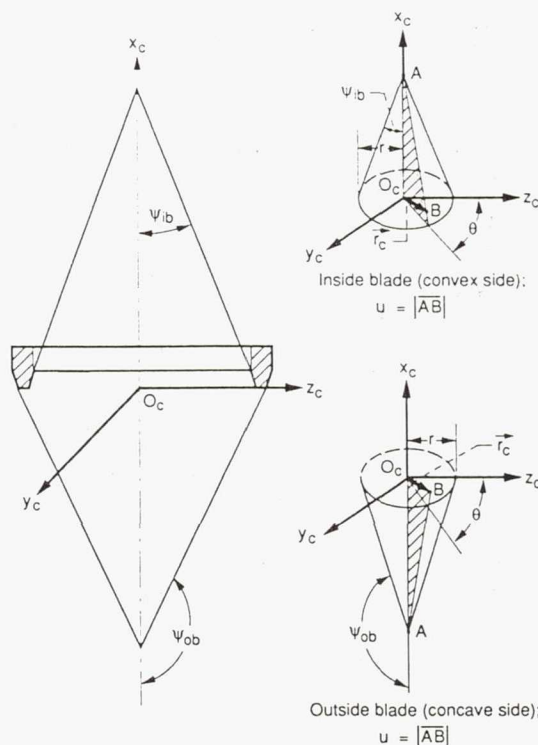


Figure 3.—Head cutter cone surfaces.

current point in the cutter coordinate system S_c . Angles ψ_{ib} and ψ_{ob} are the blades that cut the convex and concave sides of the gear tooth respectively. Thus \vec{r}_c is given by the following equation:

$$\vec{r}_c = \begin{bmatrix} r \cot \psi - u \cos \psi \\ u \sin \psi \sin \theta \\ u \sin \psi \cos \theta \\ 1 \end{bmatrix} \quad (1)$$

Once we have \vec{r}_c we then transform from one coordinate system to the next for the coordinate systems as shown in Figs. (4) and (5) [27].

Using matrix transformations we can determine the coordinates in S_w of a point on the generating surface by:

$$\vec{r}_w = [M_{wa}(f(\phi_c))] [M_{ap}] [M_{pm}] [M_{ms}(\phi_c)] [M_{sc}] \vec{r}_c(u, \theta) \quad (2)$$

Here $[M_{ij}]$ describes the required homogeneous coordinate transformation from system "j" to system "i". Therefore Eq. (2) describes the location of a point in the gear fixed coordinate system based on machine settings L_m , E_m , q , s , r , ψ , and gear design information μ and δ . At this point the machine settings and the gear design values are known. Parameters u , θ , and ϕ_c are the unknown variables that are solved for numerically.

Tooth Surface Coordinate Solution Procedure

In order to solve for the coordinates of a spiral bevel gear-tooth surface, the following

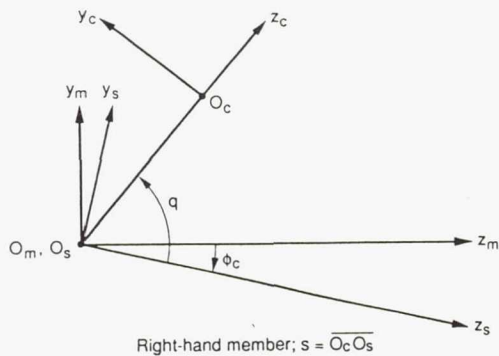
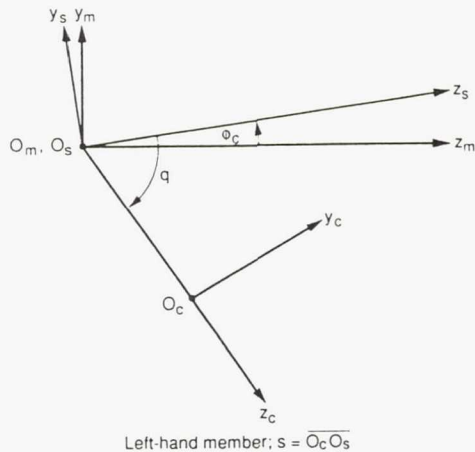


Figure 4.—Orientation of cutter, cradle, and fixed coordinate systems, S_c , S_s , and S_m respectively.

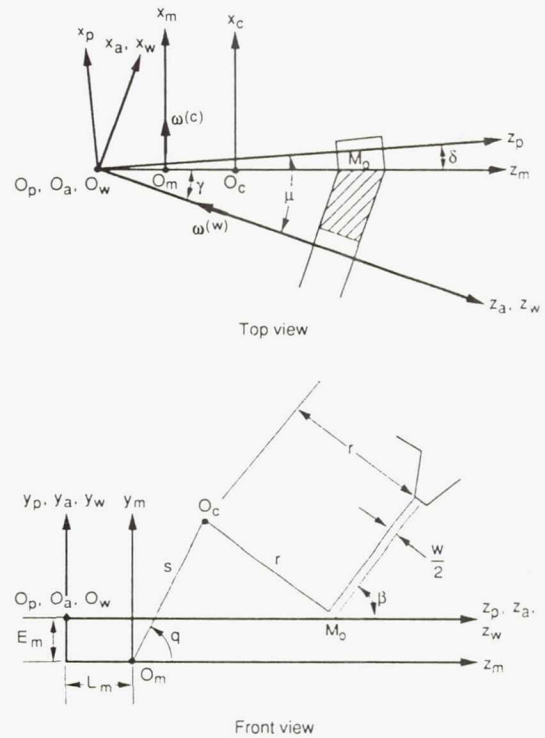


Figure 5.—Top and front view of right hand gear surface generation ($\phi_c = 0$ shown here).

items must be used simultaneously: the transformation process, the equation of meshing, and the basic gear design information. The transformation process described previously is used to determine the location of a point on the head cutter in coordinate system S_w . Since there are three unknown quantities (u , θ , and ϕ_c), three equations relating them must be developed.

Values for u , θ , and ϕ_c are used to satisfy the equation of meshing given by Refs. 1 and 9.

$$\vec{n} \cdot \vec{V} = 0 \quad (3)$$

where \vec{n} is the normal vector to the cutter and workpiece surfaces at the specified location of interest, and \vec{V} is the relative velocity between the cutter and workpiece surfaces at the specified location.

Gear design information is then used to establish an allowable range of values of the radial (\bar{r}) and axial (\bar{z}) positions that are known to exist on the gear being generated. This is shown in Fig. 6.

First the equation of meshing must be satisfied. This is given as [9]:

$$\begin{aligned} & (u - r \cot \psi \cos \psi) \cos \gamma \sin(\theta \mp q \pm \phi_c) \\ & + s \left[(\phi_c / \phi_w) - \sin \gamma \right] \cos \psi \sin \theta \mp \cos \gamma \sin \psi \sin(q - \phi_c) \\ & \pm E_m \left(\cos \gamma \sin \psi + \sin \gamma \cos \psi \cos(\theta \mp q \pm \phi_c) \right) \\ & - L_m \left(\sin \gamma \cos \psi \sin(\theta \mp q \pm \phi_c) \right) = 0 \end{aligned}$$

TABLE I. - EXAMPLE CASE FOR SURFACE COORDINATE GENERATION

Gear design data		
Number of teeth (pinion, gear)	12, 36	
Dedendum angle, δ , deg	1.0	
Addendum angle, deg	3.883	
Pitch angle, μ , deg	18.433	
Shaft angle, deg	90.0	
Mean spiral angle, deg	35.0	
Face width, mm, (in.)	25.4 (1.0)	
Mean cone distance, mm (in.)	81.05 (3.191)	
Inside radius of gear blank, mm (in.)	15.3 (0.6094)	
Top land thickness, mm (in.)	2.032 (0.080)	
Clearance mm (in.)	0.762 (0.030)	
Generation machine settings		
	Concave	Convex
Radius of cutter, r , mm (in.)	75.222 (2.9615)	78.1329 (3.0761)
Blade angle, ψ , deg	161.358	24.932
Vector sum, L_m , mm, in.	1.0363 (0.0408)	-1.4249 (-0.0561)
Machine offset, E_m , mm (in.)	3.9802 (0.1567)	-4.4856 (-0.1766)
Cradle to cutter distance, s , mm (in.)	74.839 (2.9646)	71.247 (2.8050)
Cradle angle, q , deg	64.01	53.82

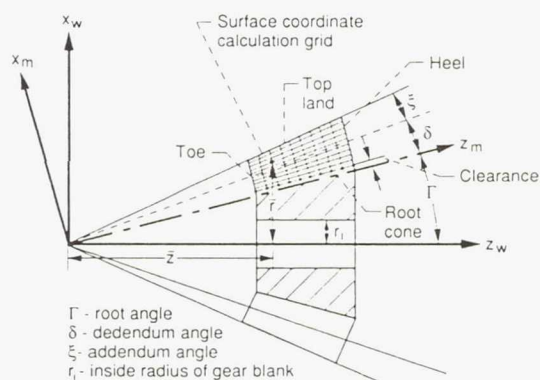


Figure 9.—Cross section of calculation grid.

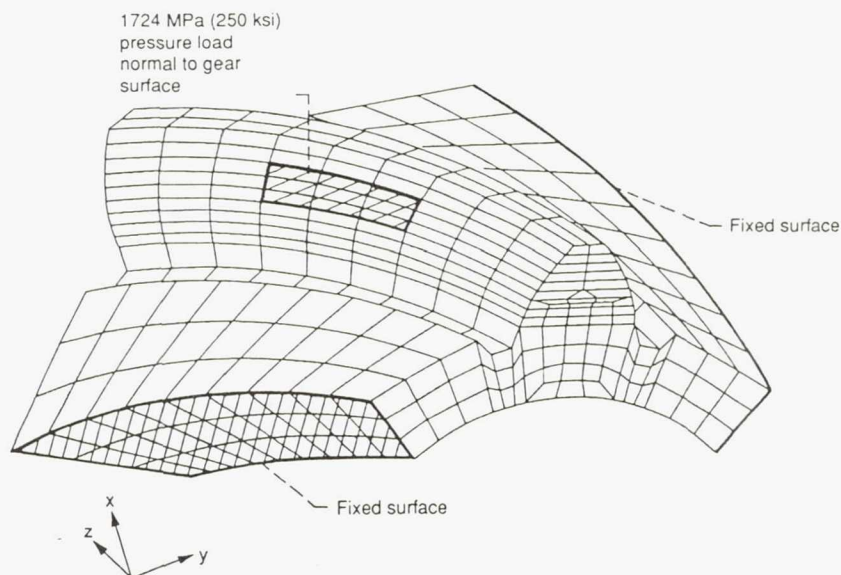


Figure 10.—Boundary conditions for the constant fillet/root radius model for the example application.

model shown in Fig. 10 is that for a constant fillet and root radius (also called full fillet) model. The fillet and root radius on the convex side has been added along with the tooth section (without the tooth) to make the model symmetric about the tooth centerline. Figure 10 shows a hidden line plot of the finite element mesh with eight-noded isoperimetric three-dimensional solid continuum elements. This model has 765 elements and 1120 nodes. The boundary conditions are also

shown in Fig. 10. A 1724-MPa (250 ksi) constant pressure load was applied normal to the tooth surface of nine elements, and the two edge surfaces of the gear rim had all degrees of freedom constrained.

The results were calculated by MSC/NASTRAN and were subsequently displayed by PATRAN. Figure 11 shows the major principle stresses for the boundary conditions shown in Fig. 10.

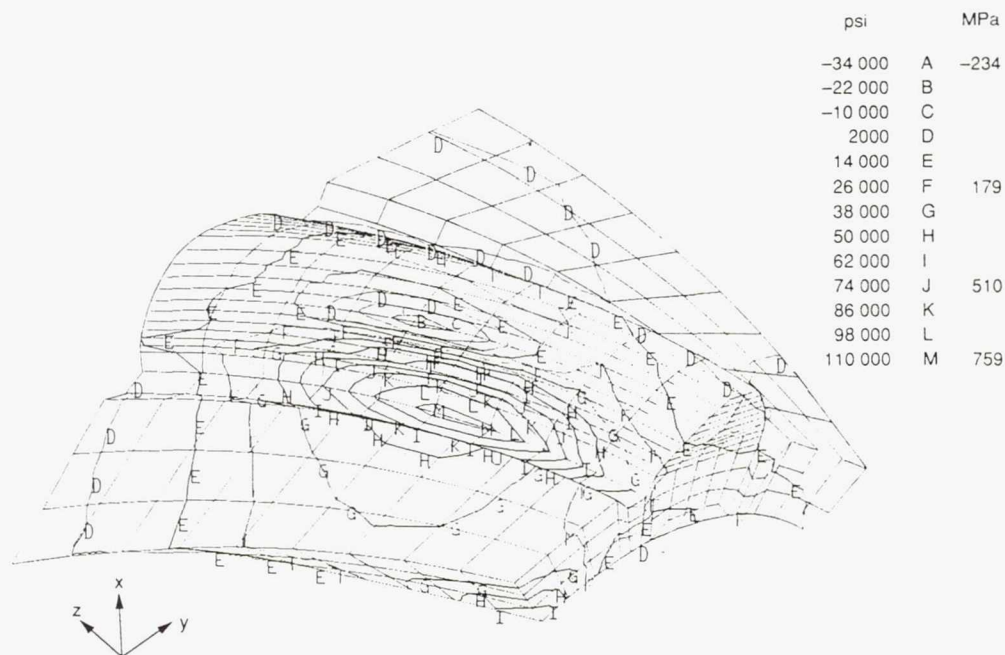


Figure 11.—Major principle stress for the boundary conditions specified in figure 10.

SUMMARY OF RESULTS

A method has been presented that uses differential geometry techniques to calculate the surface coordinates of face-milled spiral bevel gear teeth. The coordinates must be solved for numerically by a simultaneous solution of non-linear algebraic equations. These equations relate the kinematics of manufacture to the gear design parameters. Coordinates for a grid of points are determined for both the concave and convex sides of the gear tooth. These coordinates are then combined to form the enclosed surface of one gear tooth. A computer program, was developed to solve for the gear-tooth surface coordinates and provide input to a three-dimensional geometric modeling program (i.e., PATRAN). This enables an analysis by the finite element method. An example of the technique was presented.

REFERENCES

1. Litvin, F.L., "Theory of Gearing," NASA RP-1212, 1989.
2. Litvin, F.L. and Choy, J., "Spiral Bevel Geometry and Gear Train Precision." Advanced Power Transmission Technology, NASA CP-2210, AVRADCOM TR 82-C-16, G.K. Fischer, ed., 1981, pp. 335-344.
3. Huston, R. and Coy, J., "A Basis for the Analysis of Surface Geometry of Spiral Bevel Gears." Advanced Power Transmission Technology, NASA CP-2210, AVRADCOM TR 82-C-16, G.K. Fischer, ed., 1981, pp. 317-334.
4. Chang, S.H., Huston, R.L., and Coy, J.J., "Computer Aided Design of Bevel Gear Tooth Surfaces." Proceedings of the 1989 International Power Transmission and Gearing Conference, 5th, Vol. 2, ASME, 1989, pp. 585-591. (Also, NASA TM-101449).
5. Litvin, F.L., et al., "Method for Generation of Spiral Bevel Gears With Conjugate Gear Tooth Surfaces." J. Mech. Trans. Automat. Des., vol. 109, no. 2, June 1987, pp. 163-170.
6. Tsai, Y.C. and Chin, P.C., "Surface Geometry of Straight and Spiral Bevel Gears." J. Mech. trans. Automat. Des., vol. 109, no. 4, Dec. 1987, pp. 443-449.
7. Clouiter, L. and Gosselin, C., Kinematic Analysis of Bevel Gears. ASME Paper 84-DET-177, Oct. 1984.

8. Litvin, F.L., Tsung, W.J., and Lee, H.T., "Generation of Spiral Bevel Gears With Conjugate Tooth Surfaces and Tooth Contact Analysis." NASA CR-4088, AVSCOM-TR 87-C-22, 1987.
9. Litvin, F.L. and Lee, H.T., "Generation and Tooth Contact Analysis of Spiral Bevel Gears With Predesigned Parabolic Functions of Transmission Errors," NASA CR-4259, AVSCOM-TR-89-C-014, 1989.
10. Litvin, F.L., et al., "Transmission Errors and Bearing Contact of Spur, Helical and Spiral Bevel Gears." SAE Paper 881294, Sept. 1988.
11. Litvin, F.L., et al., "New Generation Methods for Spur, Helical, and Spiral Bevel Gears." NASA TM-88862, AVSCOM-TR-86-C-27, 1986.
12. Exact Determination of Tooth Surfaces for Spiral Bevel and Hypoid Gears. Gleason Works, Rochester, NY, Aug. 1970.
13. Mark, W.D., "Analysis of the Vibratory Excitation Arising From Spiral Bevel Gears." NASA CR-4081, 1987.
14. Kahraman, A., et al., "Dynamic Analysis of Geared Rotors by Finite Elements." Proceedings of the 1989 International Power Transmission and Gearing Conference, 5th, Vol. 1, ASME, 1989, pp. 375-382. (also, NASA TM-102349).
15. Drago, R. and Margasahayam, R., "Analysis of the Resonant Response of Helicopter Gears With the 3-D Finite Element Method." Presented at the 1988 MSC/NASTRAN World User's Conference, Mar. 1988.
16. Chao, H.C., Baxter, M., and Cheng, H.S., "A Computer Solution for the Dynamic Load, Lubricant Film Thickness, and Surface Temperatures in Spiral Bevel Gears." Advanced Power Transmission Technology, NASA CP-2210, AVRADCOM TR-82-C-16, G.K. Fischer, ed., 1981, pp. 345-364.
17. Winter, H. and Paul, M., "Influence of Relative Displacements Between Pinion and Gear on Tooth Root Stresses of Spiral Bevel Gears." J. Mech. Trans. Automat. Des., vol. 107, no. 1, Mar. 1985, pp. 43-48.
18. Oswald, F.B., "Gear Tooth Stress Measurements on the UH-60A Helicopter Transmission." NASA TP-2698, 1987.
19. Pizzigati, G.A. and Drago, R.J., "Some Progress in the Accurate Evaluation of Tooth Root and Fillet Stresses in Lightweight, Thin-Rimmed Gears." AGMA Paper 229.21. American Gear Manufacturers Association, Oct. 1980.
20. Wilcox, L.E., "Analyzing Gear Tooth Stress as a Function of Tooth Contact Pattern Shape and Position." Gear Technology, vol. 2, no. 1, Jan./Feb. 1985, pp. 9-15.23 (also, AGMA Paper 229.25, 1982).
21. Drago, R.J. and Uppaluri, B.R., "Large Rotorcraft Transmission Technology Development Program," Vol. 1, Technical Report. (D210-11972-1-VOL-1, Boeing Vertol Co.; NASA Contract NAS3-22143) NASA CR168116, 1983.
22. Baxter, M.L., Jr., "Effect of Misalignment on Tooth Action of Bevel and Hypoid Gears." ASME Paper 61-MD-20, Mar. 1961.
23. Spear, G.M. and Baxter, M.L., "Adjustment Characteristics of Spiral Bevel and Hypoid Gears." ASME Paper 66-MECH-17, Oct. 1966.
24. Frint, H., "Automated Inspection and Precision Grinding of Spiral Bevel Gears," NASA CR-4083, AVSCOM-TR-87-C-11, 1987.
25. Litvin, F.L., et al., "Computerized Inspection of Gear Tooth Surfaces." NASA TM-102395, AVSCOM-TR-89-C-011, 1989.
26. Lewicki, D.G. and Coy, J.J., "Vibration Characteristics of OH-58A Helicopter Main Rotor Transmission." NASA TP-2705, AVSCOM-TR-86-C-42, 1987.
27. Handschuh, R. and Litvin, F.L., "A Method for Determining Spiral Bevel Gear Tooth Geometry for Finite Element Analysis." NASA TP-3096, USAAVSCOM-TR-91-C-020, 1991.



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